



WELCOME to CAUx Local India 2018



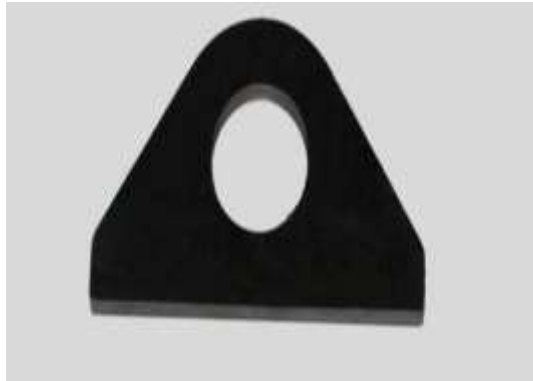
Lifting Lug Design In Detail

Prepared by Sachin Pol and Fauzan Badiwale

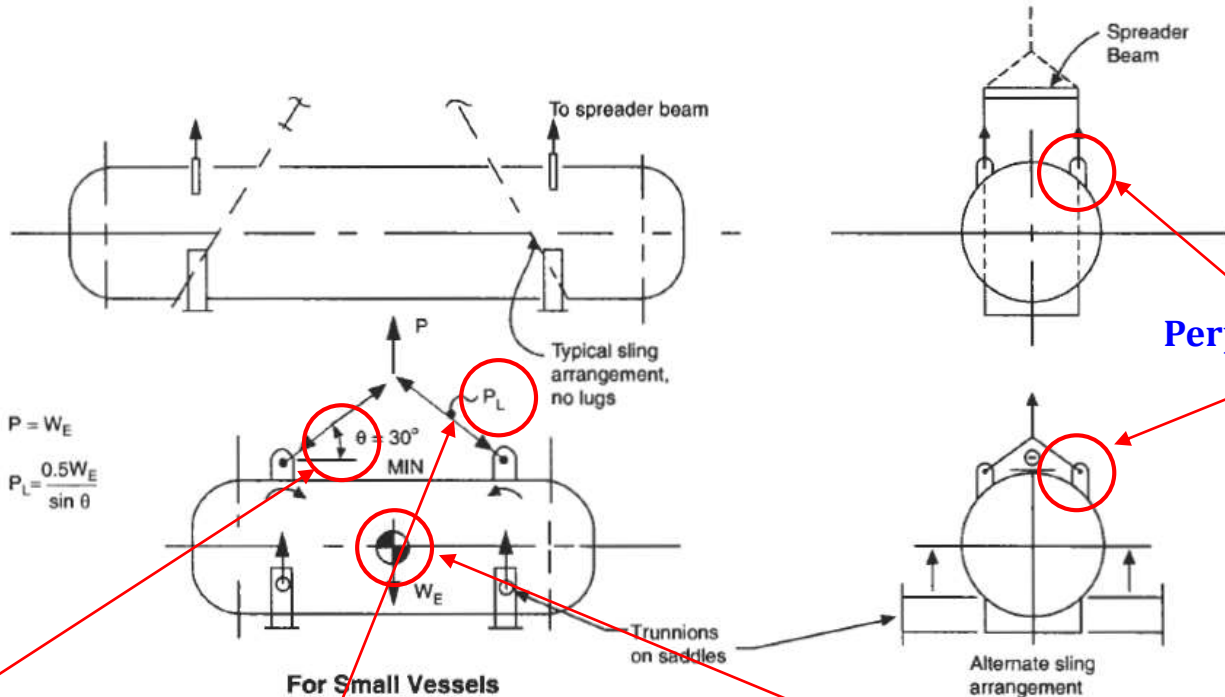
Typical Lifting Arrangements for Vessel



Different types of Lifting Arrangements



Typical Lifting Arrangements for Horizontal Vessels

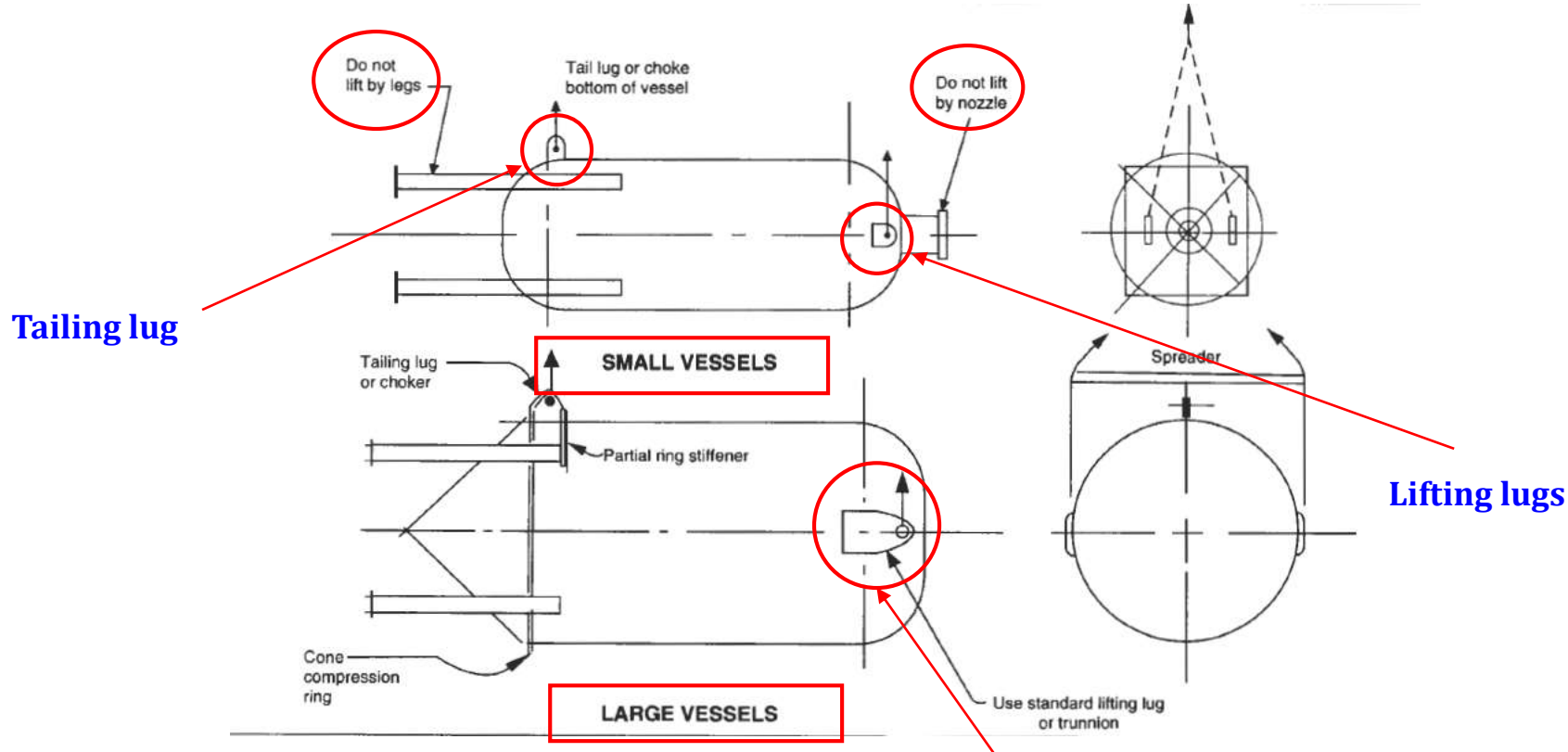


θ is Sling Angle

P_L is load acting on the sling

C.G of vessel

Typical Lifting Arrangements for Leg Supported Vessels



For large vessels Trunnion or Standard Heavy lifting lugs are used

Horizontal to vertical lifting Forces

Free-Body Diagram

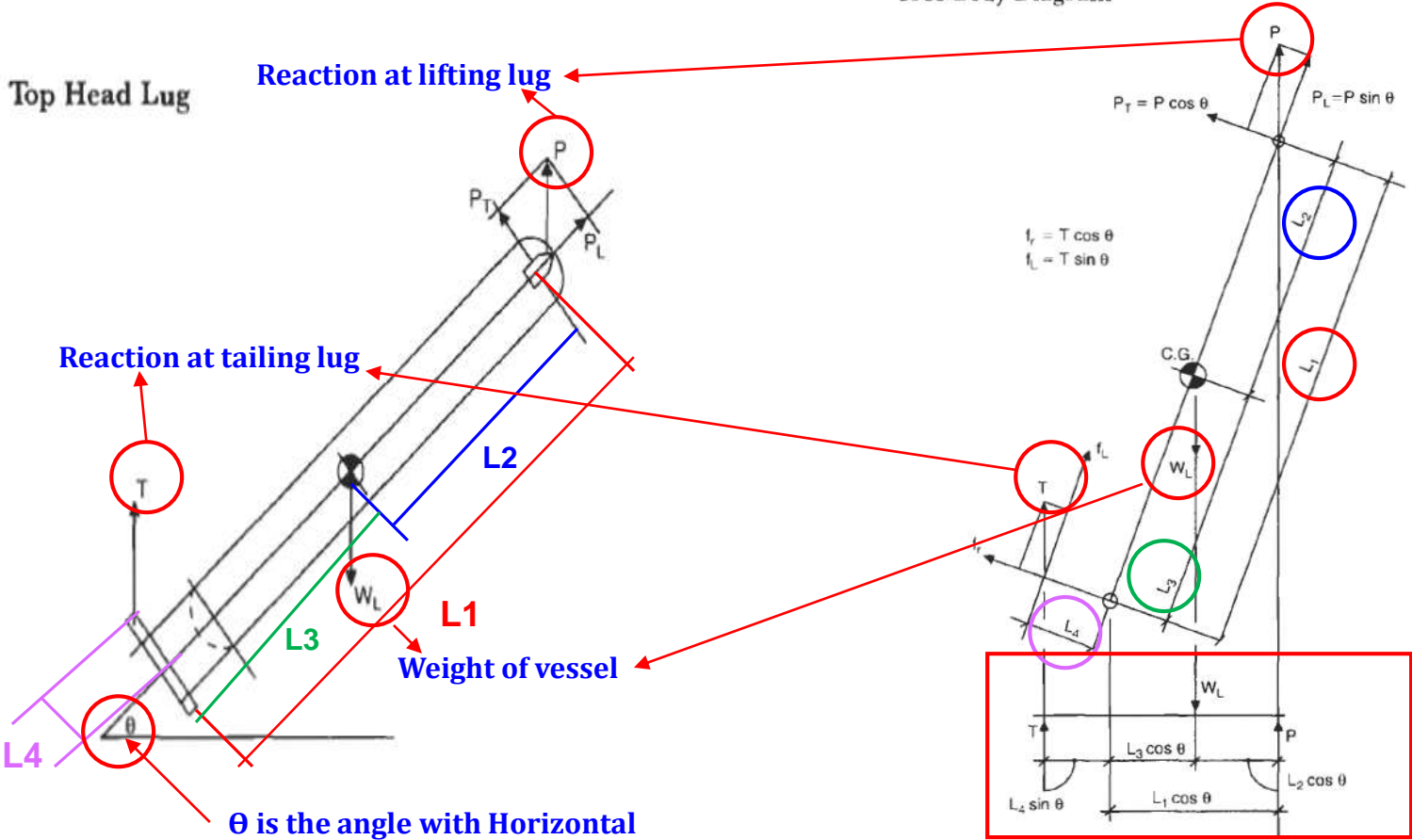
Top Head Lug

Reaction at lifting lug

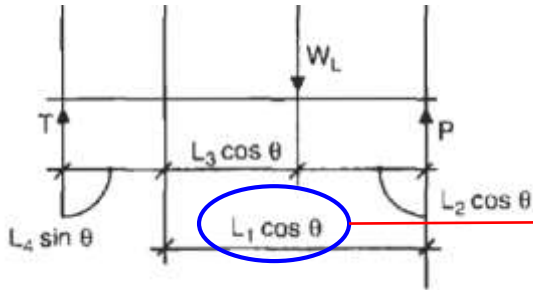
Reaction at tailing lug

Weight of vessel

θ is the angle with Horizontal



Horizontal to vertical lifting Forces Calculations



$$W_L * L_3 \cos\theta = W_L * L_1 \cos\theta - T * L_1 \cos\theta - T * L_4 \sin\theta$$

$$L_1 \cos\theta = L_2 \cos\theta + L_3 \cos\theta$$

~~$$W_L * L_3 \cos\theta = W_L * L_2 \cos\theta + W_L * L_3 \cos\theta - T * L_1 \cos\theta - T * L_4 \sin\theta$$~~

$$T * L_1 \cos\theta + T * L_4 \sin\theta = W_L * L_2 \cos\theta$$

Resolving the forces in vertical direction

$$W_L = T + P$$

Hence ,

$$P = W_L - T$$

Taking moment at T for equilibrium of the forces,

$$W_L * (L_3 \cos\theta + L_4 \sin\theta) = P * (L_1 \cos\theta + L_4 \sin\theta)$$

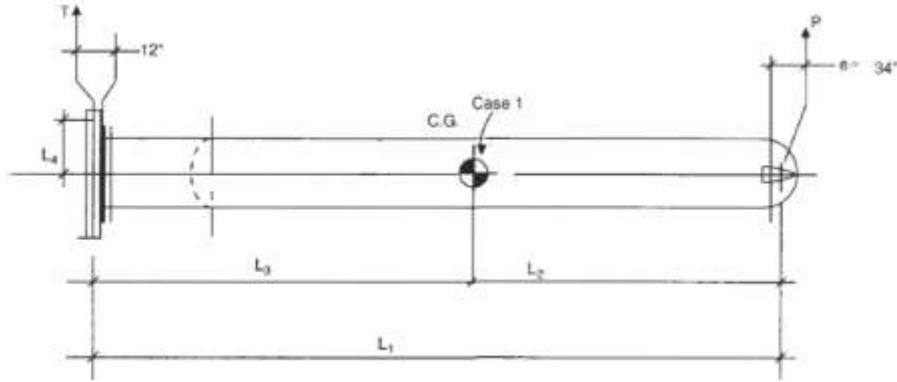
$$T = \frac{W_L * L_2 \cos\theta}{L_1 \cos\theta + L_4 \sin\theta}$$

$$W_L * (L_3 \cos\theta + L_4 \sin\theta) = (W_L - T) * (L_1 \cos\theta + L_4 \sin\theta)$$

~~$$W_L * L_3 \cos\theta + W_L * L_4 \sin\theta = W_L * L_1 \cos\theta + W_L * L_4 \sin\theta - T * L_1 \cos\theta - T * L_4 \sin\theta$$~~

Horizontal to vertical lifting Forces Calculations

Sample Problem



Where, $L_3 > L_2$

By Using Following Equations loads are calculated,

$$T = \frac{W_L * L_2 \cos\theta}{L_1 \cos\theta + L_4 \sin\theta}$$

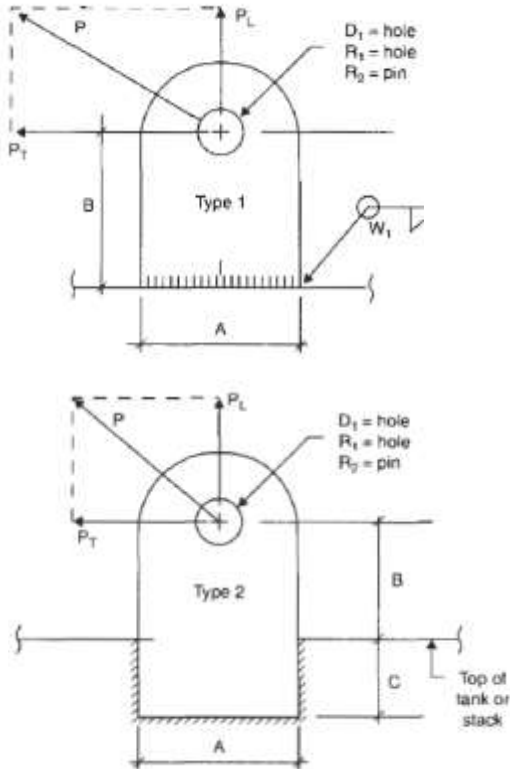
$$P = W_L - T$$

| Loads T and P | | |
|---------------|-------|-------|
| θ | T | P |
| 0 | 171.7 | 228.3 |
| 10 | 170.6 | 229.4 |
| 20 | 169.6 | 230.4 |
| 30 | 168.3 | 231.7 |
| 40 | 166.8 | 233.2 |
| 50 | 164.8 | 235.2 |
| 60 | 161.9 | 238.1 |
| 70 | 156.6 | 243.4 |
| 80 | 143.2 | 256.8 |
| 90 | 0 | 400 |



Lifting Lug Design

Thickness calculations



Thickness Due to bending = t_L

$$t_L = \frac{6P_T B}{A^2 F_b}$$

Now,

$$F_b = \frac{M}{Z}$$

$M = P_T \times B$

$Z = I / Y$

Hence, $F_b = \frac{P_T \times B}{\frac{t_L \times A^2}{6}} = \frac{6 P_T B}{t_L \times A^2}$

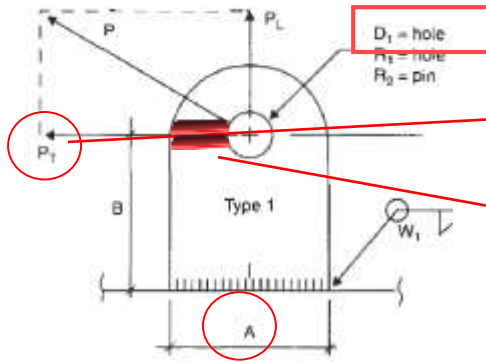
Here, $\frac{t_L \times A^3}{12 \times 6} / (A/2)$

Now, $t_L = \frac{6 P_T \times B}{F_b \times A^2}$

$$Z = \frac{t_L \times A^2}{6}$$

Lifting Lug Design

Thickness calculations

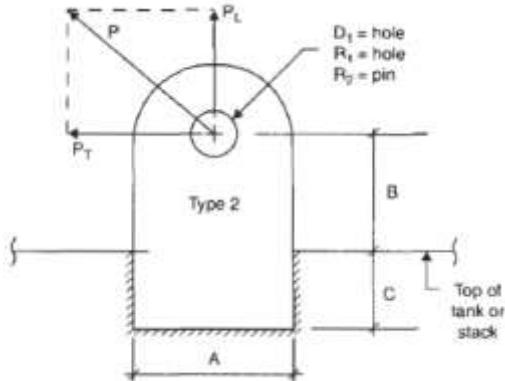


Thickness Due to Shear = t_L

$$t_L = \frac{P_T}{(A - D_1)F_s}$$

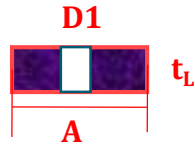
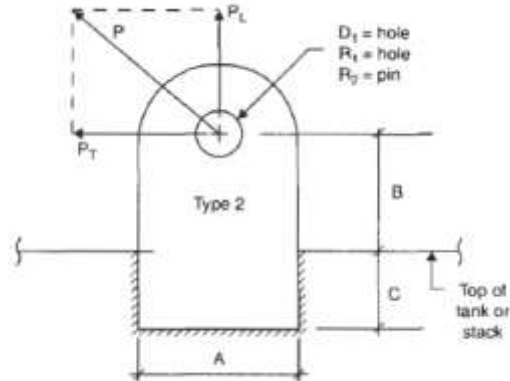
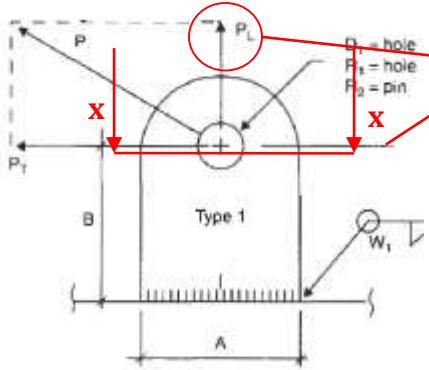
$$\text{Shear Stress } F_s = \frac{\text{Shear Force}}{\text{Shear Area}} = \frac{P_T}{\frac{A - D_1}{2} \times 2 \times t_L}$$

$$\text{Hence, } t_L = \frac{P_T}{(A - D_1) \times F_s}$$



Lifting Lug Design

Thickness calculations



Thickness Due to Tension = t_L

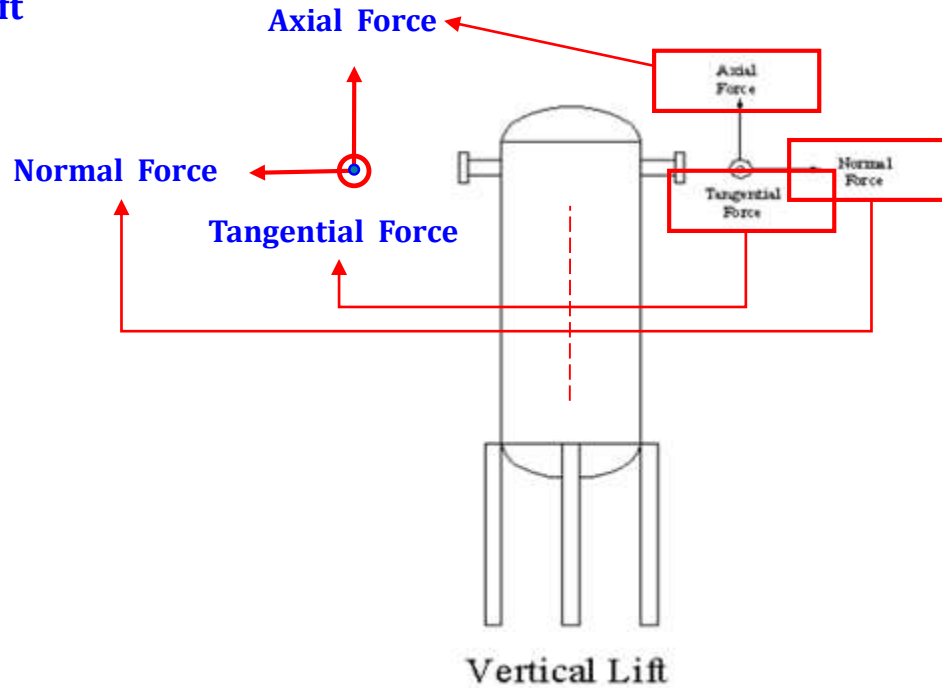
$$t_L = \frac{P_L}{(A - D_1)F_t}$$

$$\text{Tensile Stress } F_t = \frac{\text{Tensile Force}}{\text{Area in tension}} = \frac{P_L}{(A - D_1) \times t_L}$$

$$\text{Hence, } t_L = \frac{P_L}{(A - D_1) \times F_t}$$

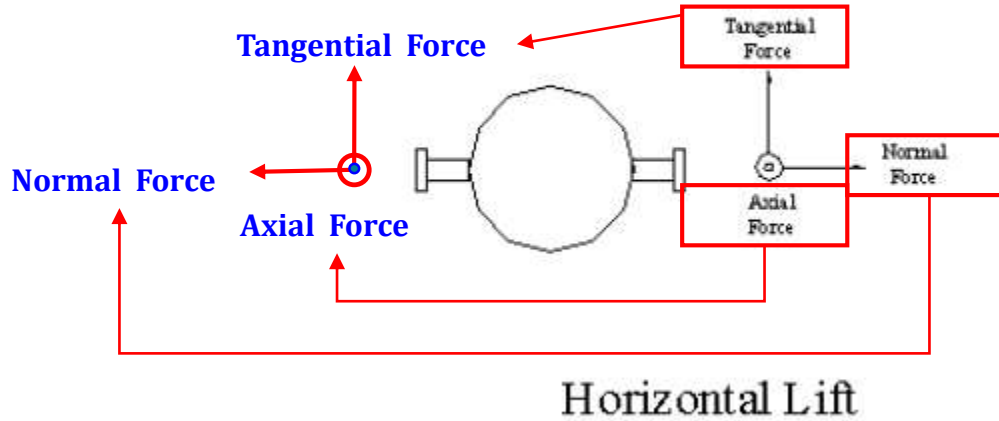
PV Elite Forces and sign Conventions

For Vertical Lift

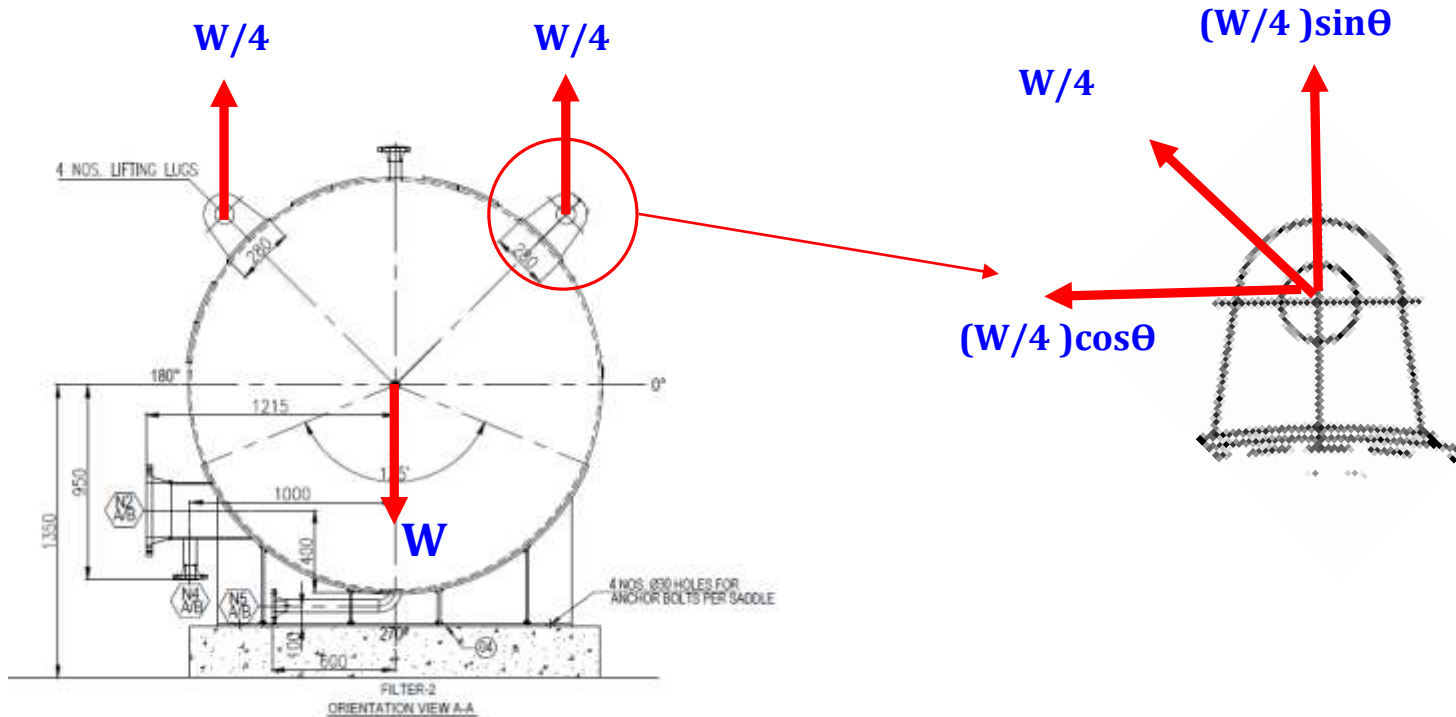


PV Elite Forces and sign Conventions

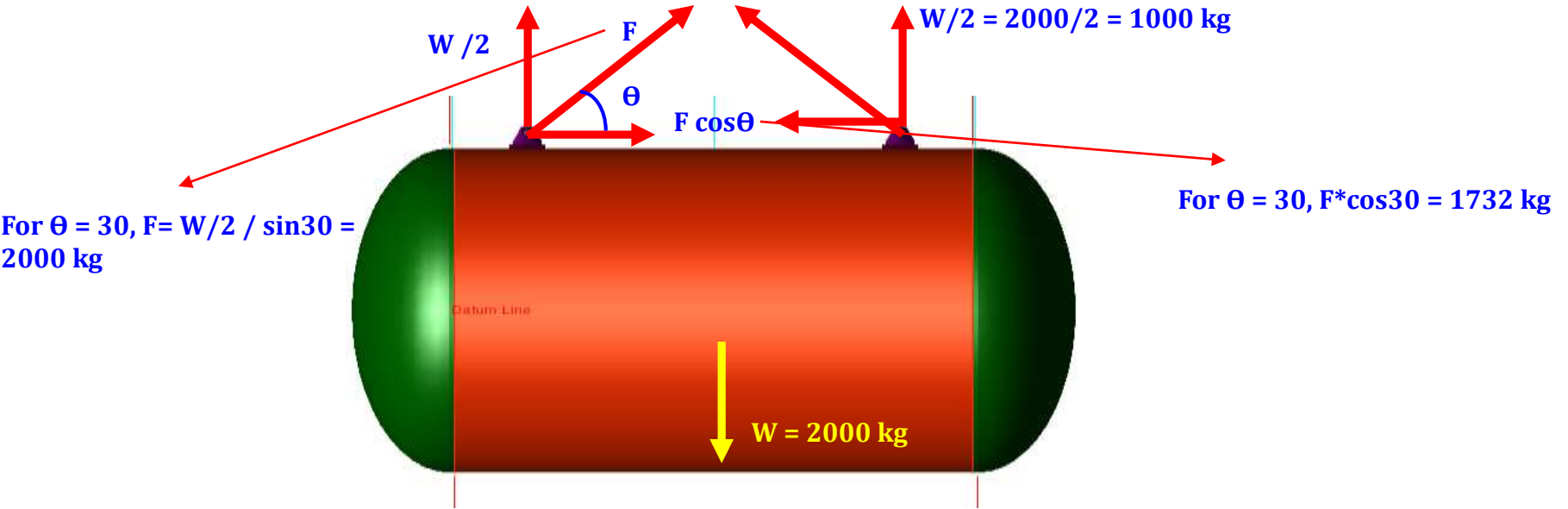
For Horizontal Lift



PV Elite Forces and sign Conventions



PV Elite Lifting Lug Sample Example



Vessel I.D = 1000 mm , Shell Thickness = 6mm , Weight = 2000 kg ,

2 Nos of perpendicular lifting lugs provided

PV Elite Lifting Lug Sample Example

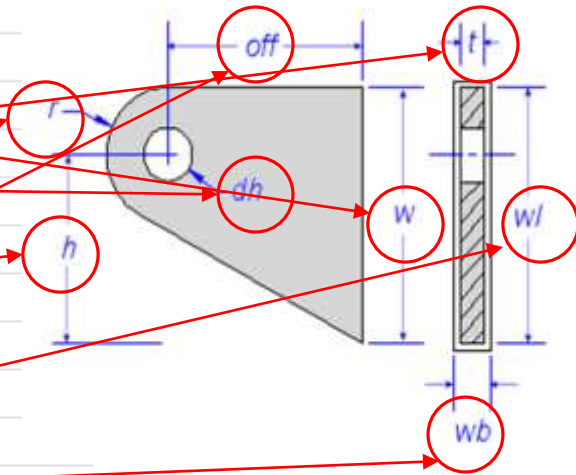
| Identification | |
|---|--------------------------|
| Item Number | 1 |
| Description | Lifting Lug |
| Legs and Lugs | |
| Design Pressure, kgf/cm ² | 3.5 |
| Design Temperature for Internal Pressure, C | 85 |
| Outside Diameter of Vessel, mm | 1012 |
| Shell Thickness, mm | 6 |
| Shell Corrosion Allowance, mm | 0 |
| Tangent to Tangent Length of Vessel, cm | 420 |
| Shell Material | SA-516 70 |
| Type of Analysis | Lifting Lug |
| Analyze Baseplate ? | <input type="checkbox"/> |

| | |
|--|--------------------------|
| Additional Horizontal Force on Vessel, kgf | 0 |
| Location of Horizontal Force above Base Point, cm | 0 |
| Empty Weight of Vessel, kgf | 2000 |
| Operating Weight of Vessel (total vertical load), kg | 0 |
| Height of Bottom Tangent above Base Point, cm | 0 |
| Occasional Load Factor (AISC A5.2) | 1 |
| Apply Wind Loads to Vessel ? | <input type="checkbox"/> |
| Apply Seismic Loads to Vessel ? | <input type="checkbox"/> |

PV Elite Lifting Lug Sample Example

▾ Lifting Lug

| | |
|--|---------------|
| ▾ Lifting Lug Material | SA-516 70 |
| Lug Orientation to Vessel | Perpendicular |
| Contact Width or Height (Perp. Lug) of Lifting Lug [w], mm | 150 |
| Thickness of Lifting Lug [t], mm | 16 |
| Diameter of Hole in Lifting Lug [dh], mm | 50 |
| Radius of Semi-circular Arc of Lifting Lug [r], mm | 60 |
| Height of the Lug from Bottom to Center of Hole [h], mm | 75 |
| Offset from Vessel OD to Center of Hole [off], mm | 100 |
| Minimum thickness of Fillet Weld around Lug, mm | 6 |
| Length of weld along side of Lifting Lug [wl], mm | 150 |
| Length of weld along bottom of Lifting Lug [wb], mm | 28 |



▾ Lift Information and Loads on one Lug

| | |
|-----------------------------|-------------|
| Lift Orientation (optional) | Horizontal |
| Axial Force, kgf | 1732 |
| Normal Force, kgf | 1000 |
| Tangential Force, kgf | 0 |
| Impact Factor | 1.5 |

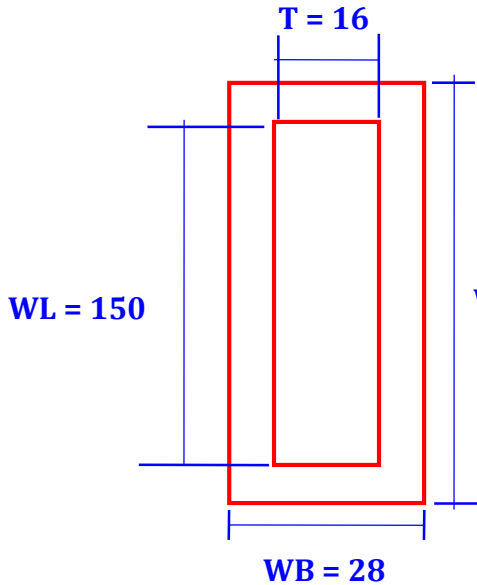
PV Elite Lifting Lug Sample Example

Results for lifting lugs, Description : Lifting Lug

Weld Group Inertia about the Circumferential Axis ILC
 Weld Group Centroid distance in the Long. Direction YLL
 Dist. of Weld Group Centroid from Lug bottom YLL_B
 Weld Group Inertia about the Longitudinal Axis ILL
 Weld Group Centroid Distance in the Circ. Direction YLC

542.023 cm**4
 81.000 mm
 75.000 mm
 24.515 cm**4
 14.000 mm

$$\frac{162}{2}$$



$$\text{Moment of Inertia ILC} = \frac{2.8 \times 16.2^3 - 1.6 \times 15^3}{12} = 542.023$$

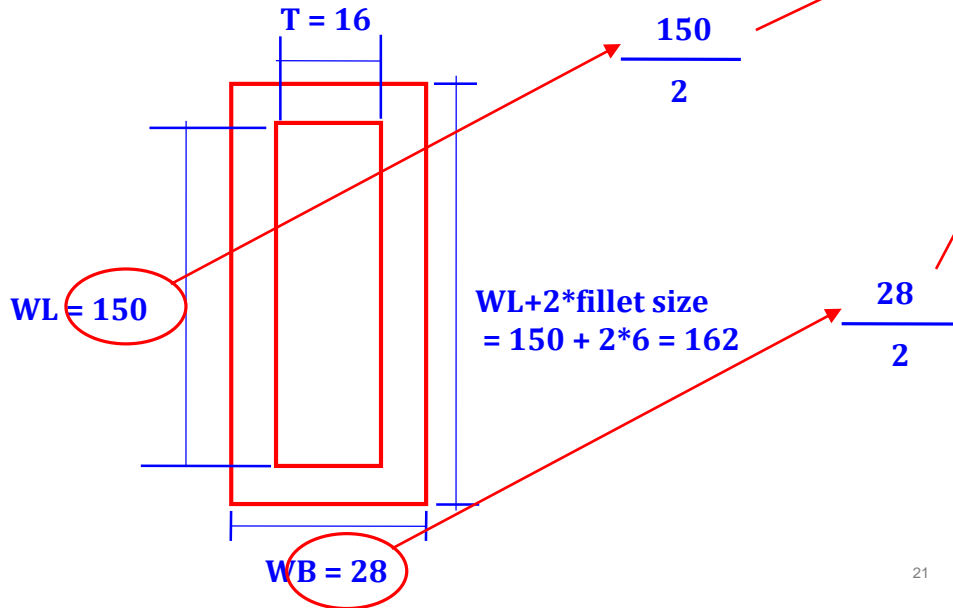
$$\text{Moment of Inertia ILL} = \frac{2.8^3 \times 16.2 - 1.6^3 \times 15}{12} = 24.515$$

$$\text{WL} + 2 \times \text{fillet size} = 150 + 2 \times 6 = 162$$

PV Elite Lifting Lug Sample Example

Results for lifting lugs, Description : Lifting Lug

| | | | |
|---|-------|---------|-------|
| Weld Group Inertia about the Circumferential Axis | ILC | 542.023 | cm**4 |
| Weld Group Centroid distance in the Long. Direction | YLL | 81.000 | mm |
| Dist. of Weld Group Centroid from Lug bottom | YLL_B | 75.000 | mm |
| Weld Group Inertia about the Longitudinal Axis | ILL | 24.515 | cm**4 |
| Weld Group Centroid Distance in the Circ. Direction | YLC | 14.000 | mm |



PV Elite Lifting Lug Sample Example

Applying the Impact factor to the loads:

$$F_{ax} = 1732.00 * 1.50 = 2598.00 \text{ kgf}$$

$$F_n = 1000.00 * 1.50 = 1500.00 \text{ kgf}$$

Primary Shear Stress in the Welds due to Shear Loads [SsII]:

$$\begin{aligned} &= \sqrt{F_{ax}^2 + F_t^2 + F_n^2} / ((2 * w_l + w_b) * t_w) \\ &= \sqrt{2598^2 + 0^2 + 1500^2} / ((2 * 150.0 + 28.0) * 6.0000) \\ &= 152.44 \text{ kgf/cm}^2 \end{aligned}$$

Total Resultant force

$$\sqrt{F_{ax}^2 + F_t^2 + F_n^2}$$

Total resisting area of weld

$$2 * (W_l + W_b) * \text{Weld size}$$

PV Elite Lifting Lug Sample Example

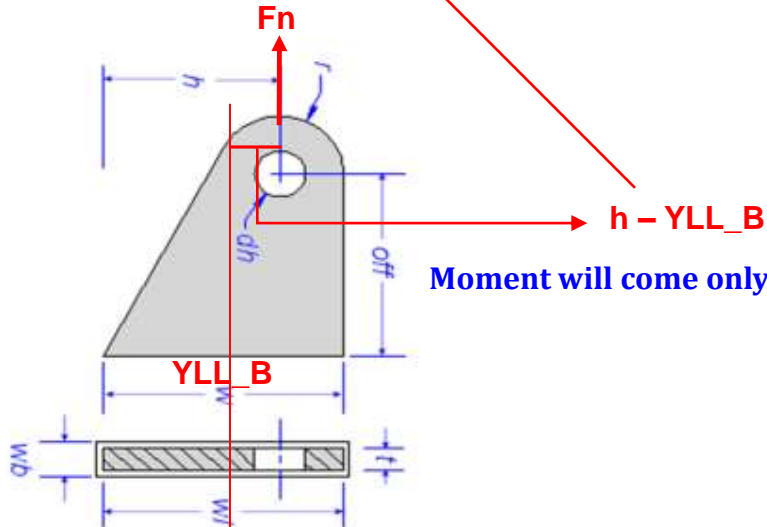
Shear Stress in the Welds due to Bending Loads [Sblf]:

$$\begin{aligned}
 &= (F_n \cdot (h - YLL_B) \cdot YLL / ILC) + (F_{ax} \cdot off \cdot YLL / ILC) + (F_t \cdot off \cdot YLC / ILL) \\
 &= (1500 \cdot (75.00 - 75.000) \cdot 81.000 / 542.023) + \\
 &\quad (1299 \cdot 100.000 \cdot 81.000 / 542.023) + \\
 &\quad (0 \cdot 100.000 \cdot 14.000 / 24.515) \\
 &= 194.13 \text{ kgf/cm}^2
 \end{aligned}$$

$$\text{Stress} = M / Z = F_n \times (h - YLL_B) / ILC / YLL = F_n \times (h - YLL_B) \cdot YLL / ILC$$

$$M = \text{Force} \cdot \text{Moment arm} = F_n \times (h - YLL_B)$$

$$Z = \text{Section modulus} = I / Y = ILC / YLL$$



Moment will come only when there is offset distance

PV Elite Lifting Lug Sample Example

Pin Hole Bearing Stress [Pbs]:

$$\begin{aligned} &= \text{Sqrt}(F_{ax}^2 + F_n^2) / (t * d_h) \\ &= \text{Sqrt}(1299^2 + 1500^2) / (16.000 * 50.000) \\ &= 248.04 \text{ kgf/cm}^2 \end{aligned}$$

Resultant Force in the sling

Bearing Area = diameter of hole x thickness of lug
= $t \times d_h$

PV Elite Lifting Lug Sample Example

Bending stress at the base of the lug [Fbs]:

$$= \frac{F_t \cdot \text{off}}{(w \cdot t^2) / 6} + \frac{F_{ax} \cdot \text{off}}{(w^2) \cdot t / 6}$$

$$= \frac{0 \cdot 100.000}{(150.000 \cdot 16.000^2) / 6} + \frac{1299 \cdot 100.000}{(150.000^2) \cdot 16.000 / 6}$$

$$= 216.51 \text{ kgf/cm}^2$$

Section modulus in strong axis

Bending moment because of Fax

Tensile stress at the base of the lug [Fa]:

$$= \frac{F_n}{w \cdot t} = 1500 / (150.000 \cdot 16.000)$$

$$= 62.50 \text{ kgf/cm}^2$$

Section modulus in weak axis

Total Combined Stress at the base of the lug:

$$= F_{bs} + F_a = 279.0 \text{ kgf/cm}^2$$

Bending moment because of Ft

Tensile force acting on lug

Cross section area at the base = w x t



Thank You!

Have a great conversation!